## Large Eddy Simulation in a Split Form Discontinuous Galerkin Method for the Compressible Navier-Stokes

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- Turbulence is prevalent in every day life
  - Most common engineering flow
- Characterized by:
  - Mechanical mixing
  - Vorticies
  - Chaotic fluctuations
  - Cascade of Energy from large to small scales
- These characteristics are a challenge to simulate





- An accurate simulation must capture all temporal and spatial scales
  - Computationally expensive
- Large Eddy Simulation
  - Reduced computational cost
  - Without sacrificing accuracy
  - Two LES models were analyzed
    - Constant Smagorinsky
    - Dynamic Smagorinsky







- High order finite element methods (FEM) have grown in popularity
  - FEM can leverage new architectural advances
  - Historically neglected due to high computational costs
- The Discontinuous Galerkin Method is a finite element method with discontinuous values at each element interface
  - Relies on the weak form of the governing equations





- An alternative DG formulation exists that relies on the strong form
  - The split-form is derived from the strong form DG
  - This split-form is kinetic energy preserving
- The behavior of the LES models when used with the split-form DG formulation were analyzed and compared to the standard DG formulation





# Direct Numerical Simulation (DNS)

- All time and space scales are simulated
- Very fine mesh resolution required
- Very small time steps required
- Very Expensive
  - Everything is directly simulatied
  - Increasing Reynolds Number increases the cost





Reynolds Averaged Navier-Stokes (RANS)

- Average flow field is calculated
- Models the fluctuations
- Much cheaper due to the reduced resolution needed
- Struggles with unsteady flow problems
- Reduction of accuracy due to model limitations
- Models need to be selected correctly for a given problem

![](_page_6_Picture_8.jpeg)

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# Large Eddy Simulation (LES)

- Directly simulate large scale structures
- Filter smallest scales (sub-grid scales)
- Introduce model for the SGS
- Cheaper than DNS, lower accuracy
- More expensive than RANS, more accuracy

![](_page_7_Picture_7.jpeg)

![](_page_8_Picture_0.jpeg)

#### Compressible Navier-Stokes

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$$\frac{\partial \boldsymbol{Q}(x,t)}{\partial t} + \vec{\nabla} \cdot \boldsymbol{F}(\boldsymbol{Q}(x,t)) = 0$$

![](_page_8_Figure_4.jpeg)

$$q_i = -\frac{C_p \mu}{Pr} \frac{\partial T}{\partial x_i}$$

![](_page_8_Picture_6.jpeg)

![](_page_9_Picture_0.jpeg)

### Weak Form

- Obtained by multiplying by a test function  $\phi_s, s = 1, ..., M$
- And integrating over the element volume

$$\int_{\Omega_k} \left( \frac{\partial \boldsymbol{Q}}{\partial t} + \vec{\nabla} \cdot \boldsymbol{F} \right) \phi(x) \, dx$$

![](_page_9_Picture_5.jpeg)

![](_page_9_Picture_6.jpeg)

![](_page_10_Picture_0.jpeg)

## Weak Form

• Separating and applying the divergence theorem

$$R^{weak} = \int_{\Omega_k} \frac{\partial Q}{\partial t} \phi(x) dx$$
$$- \int_{\Omega_k} (\mathbf{F} \cdot \vec{\nabla}) \phi(x) dx$$
$$+ \int_{\Gamma_k} (\mathbf{F}^* \cdot \vec{n}) \phi(x|_{\Gamma_k}) d\Gamma_k = 0$$

![](_page_10_Picture_4.jpeg)

![](_page_10_Picture_5.jpeg)

![](_page_11_Picture_0.jpeg)

# Strong Form

- The divergence theorem is applied a second time to the volume term
- The derivative of the volume fluxes must now be calculated
- An additional term now must be evaluated at the boundary

![](_page_11_Picture_5.jpeg)

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# Strong Form

$$R^{strong} = \int_{\Omega_k} \frac{\partial \mathbf{Q}}{\partial t} \phi(x) dx$$
$$- \int_{\Omega_k} (\vec{\nabla} \mathbf{F} \cdot \phi(x)) dx$$
$$+ \int_{\Gamma_k} ((\mathbf{F}^* - \mathbf{F}) \cdot \vec{n}) \phi(x|_{\Gamma_k}) d\Gamma_k = 0$$

![](_page_12_Picture_3.jpeg)

![](_page_13_Picture_0.jpeg)

Split-Form

- The summation-by-parts property is applied to the surface fluxes on of the strong form discretization
- Requires a coordinate transform of the generalized governing equations:

$$J\frac{\partial \widetilde{\mathbf{Q}}}{\partial t} + \widetilde{\vec{\mathcal{L}}}_X(\mathbf{Q}) + \widetilde{\vec{\mathcal{L}}}_Y(\mathbf{Q}) + \widetilde{\vec{\mathcal{L}}}_Z(\mathbf{Q}) = 0$$

![](_page_13_Picture_5.jpeg)

![](_page_14_Picture_0.jpeg)

Split-Form

 The split-form discretization is derived from the DG spectral element formulation (DGSEM) and is constructed in a similar manner to the standard DG formulation

$$\big(\widetilde{\vec{\mathcal{L}}}_X\big)_{i,j,k} \approx \frac{1}{\omega_i} \big(\delta_{iN} \big[\widetilde{\mathcal{F}}^* - \widetilde{\mathcal{F}}\big]_{Njk} - \delta i 1 \big[\widetilde{\mathcal{F}}^* - \widetilde{\mathcal{F}}\big]_{1jk} + \sum_{m=1}^N \mathbf{D}_{im}(\widetilde{\mathcal{F}})_{mjk}$$

![](_page_14_Picture_4.jpeg)

![](_page_14_Picture_5.jpeg)

![](_page_15_Picture_0.jpeg)

# Pirozzoli Numerical Flux Scheme

- A split-form discretization only flux scheme
  - Primarily used with no artificial dissipation in this work

$$\tilde{\mathcal{L}}_{X}^{PZ}(Q) = \begin{bmatrix} \frac{1}{2}((\rho u)_{x}\rho(u)_{x} + u(\rho)_{x}) \\ \frac{1}{4}((\rho u^{2})_{x} + \rho(u^{2})_{x} + 2u(\rho u)_{x} + u^{2}(\rho)_{x} + 2\rho u(u)_{x}) + p_{x} \\ \frac{1}{4}((\rho uv)_{x} + \rho(uv)_{x} + u(\rho v)_{x} + v(\rho u)_{x} + uv(\rho)_{x} + \rho v(u)_{x} + \rho u(v)_{x}) \\ \frac{1}{4}((\rho uw)_{x} + \rho(uw)_{x} + u(\rho w)_{x} + w(\rho u)_{x} + uw(\rho)_{x} + \rho w(u)_{x} + \rho u(w)_{x}) \\ \frac{1}{4}((\rho uh)_{x} + \rho(uh)_{x} + h(\rho u)_{x} + u(\rho h)_{x} + uh(\rho)_{x} + \rho u(h)_{x} + \rho u(u)_{x}) \end{bmatrix}$$

![](_page_15_Picture_5.jpeg)

![](_page_16_Picture_0.jpeg)

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# Solution Filtering

- A method for stabilizing DG methods
- Solution is filtered at each time step
  - Also any time stepping stages like in the Runge-Kutta Method
- Solution filter is also used in the Dynamic Smagorinsky Method
  - Modal Cutoff
  - Laplacian Filter

![](_page_16_Picture_8.jpeg)

![](_page_17_Picture_0.jpeg)

# Modal Cutoff Filter

- An Nth order hierarchical basis function contains all lower solutions
- Specific orders can be filtered by zeroing corresponding modes
- Emulates a sharp cut off filter
- Problem: CartDG uses a nodal basis for solution

![](_page_17_Picture_6.jpeg)

![](_page_18_Picture_0.jpeg)

# Hierarchical Modal basis

- Legendre Polynomial used to calculate
- P6 Basis shown
- All lower order basis functions are represented in the basis

![](_page_18_Figure_5.jpeg)

![](_page_18_Picture_6.jpeg)

![](_page_19_Picture_0.jpeg)

#### Nodal Basis

- Gives rise to Kronecker Delta Property used to speed up CartDG
- Each order has a unique set of basis functions
- P6 Basis shown

![](_page_19_Figure_5.jpeg)

![](_page_19_Picture_6.jpeg)

![](_page_20_Picture_0.jpeg)

## Modal Cutoff Filter

- Can be transformed by use of mass matricies
- Modal Mass Matrix

$$M_{ij} = \int_{-1}^{1} \psi_i(\xi) \psi_j(\xi) d\xi$$

• Mix Mass Matrix

$$C_{ij} = \int_{-1}^{1} \psi_i(\xi) \phi_j(\xi) d\xi$$

• This can be used to calculate

$$C\vec{u} = M\vec{b}$$
$$\vec{b} = M^{-1}C\vec{u}$$

• The filter matrix **F** can be applied

$$\overline{\vec{b}} = F\vec{b}$$

![](_page_20_Picture_11.jpeg)

![](_page_21_Picture_0.jpeg)

#### Modal Cutoff Filter

- This can be used to get the filtered solution  $\overline{\vec{u}} = C^{-1}M\overline{\vec{b}}$   $\overline{\vec{u}} = C^{-1}MFM^{-1}C\vec{u}$
- Terms can be combined into an overall filter  $\widehat{F}$

#### $\overline{\vec{u}} = \widehat{F}\vec{u}$

• Filter  $\hat{F}$  only needs to be calculated once

![](_page_21_Picture_6.jpeg)

![](_page_22_Picture_0.jpeg)

#### 1-D Test Problem

- P6 Nodal approximation in Red
- N = 6 mesh elements (blue)
- Quadrature points shown in violet

(X) 0  $^{-1}$ -2 -3 2 3 -2 1

Nodal Solution

• Black is true solution  $y = \cos(2x) + 0.3\sin(8x) + \sin(x^2) + 0.4\sin\left(\frac{36}{\pi}x\right)$ 

![](_page_22_Picture_7.jpeg)

![](_page_23_Picture_0.jpeg)

#### 1-D Test Problem: Modal Cutoff Filter

- P4 filtered solution in green
- Steep peaks are smoothed out in several areas
- Some larger discontinuities near element boundary
- High order content is removed
- Solution is slightly less accurate

![](_page_23_Figure_7.jpeg)

![](_page_23_Picture_8.jpeg)

![](_page_24_Picture_0.jpeg)

#### 1-D Test Problem: Modal Cutoff Filter

- P1 filtered solution in green
- All high order frequencies are smoothed out
- Solution accuracy is poor
- Implies that too low of filter order worsens accuracy and potentially stability

![](_page_24_Figure_6.jpeg)

![](_page_24_Picture_7.jpeg)

![](_page_25_Picture_0.jpeg)

#### 3-D Test Problem

- P6 Nodal approximation for x-momentum pu
- Taylor Green Vortex
  - At  $\tau \approx 10$
- N = 10 mesh elements in each direction
- Red is positive values of pu, green is zero, and blue is negative values

![](_page_25_Figure_7.jpeg)

![](_page_25_Picture_8.jpeg)

![](_page_26_Picture_0.jpeg)

#### 3-D Test Problem: Modal Cutoff Filter

- P4 filtered solution
- Sharp flow features smoothed out on XZface near Z = 0
- Majors structures on XZface have become more defined

![](_page_26_Figure_5.jpeg)

![](_page_26_Picture_6.jpeg)

![](_page_26_Picture_7.jpeg)

![](_page_27_Picture_0.jpeg)

#### 3-D Test Problem: Modal Cutoff Filter

- P1 filtered solution
- Structures are coarse
- Other structures smoothed out entirely
- Discontinuities from finite representation are apparent
- Suggest that too low of a filter order leads to a significant degradation in accuracy

![](_page_27_Figure_7.jpeg)

![](_page_27_Picture_8.jpeg)

![](_page_28_Picture_0.jpeg)

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## Laplace Filter

- Regularizations of the convective term in the Navier-Stokes equations
  - This nonlinear term leads to the small scale structures for the turbulent cascade
- The regularization of this term can lead to the convection term becoming a source or a sink
  - This can be corrected by projecting onto a divergence free space
- This lead to this work investigating the Laplace Filter in the dynamic Smagorinsky model

![](_page_28_Picture_7.jpeg)

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#### Laplace Filter

• Calculated explicitly:

$$\overline{u} = u + \nabla \cdot (\gamma \nabla u)$$

- Where the filtered solution  $\bar{u}$  is filtered based on the divergence of the flow
- $\gamma$  is a filter width term
  - Box filter was selected on a per element basis

$$\gamma = \frac{\left(\frac{1}{p+1} \times (\Omega_k)^{\frac{1}{3}}\right)^2}{24}$$

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- With the element volume  $\Omega_{\textbf{k}}$  normalized by the solution order p

![](_page_29_Picture_9.jpeg)

![](_page_30_Picture_0.jpeg)

#### 1-D Test Problem

- P6 Nodal approximation in Red
- N = 6 mesh elements (blue)
- Quadrature points shown in violet

(X) 0  $^{-1}$ -2 -3 2 3 -2 1

Nodal Solution

• Black is true solution  $y = \cos(2x) + 0.3\sin(8x) + \sin(x^2) + 0.4\sin\left(\frac{36}{\pi}x\right)$ 

![](_page_30_Picture_7.jpeg)

![](_page_31_Picture_0.jpeg)

## 1-D Test Problem: Laplace Filter

- Laplacian filtered solution in green
- Minimal changes in filtered solution
  - Derivative of a sinusoid is a sinusoid
- Slight changes near the edges of each element

![](_page_31_Figure_6.jpeg)

![](_page_31_Picture_7.jpeg)

![](_page_32_Picture_0.jpeg)

#### 3-D Test Problem

- P6 Nodal approximation for x-momentum pu
- Taylor Green Vortex
  - At  $\tau \approx 10$
- N = 10 mesh elements in each direction
- Red is positive values of pu, green is zero, and blue is negative values

![](_page_32_Figure_7.jpeg)

![](_page_32_Picture_8.jpeg)

![](_page_33_Picture_0.jpeg)

#### 3-D Test Problem: Laplace Filter

- Laplace filtered solution
- Formerly smooth areas now have structures
  - These arise from small changes in the sign of the solution
- Areas with more gradual changes are smoothed out
- This makes it ideal for the filter used in LES
  - These areas are more likely to be under resolved

![](_page_33_Figure_8.jpeg)

![](_page_33_Picture_9.jpeg)

![](_page_33_Picture_10.jpeg)

![](_page_34_Picture_0.jpeg)

# LES Equations

- To obtain these equations a low-pass filter is applied to the Navier-Stokes equations
  - Applied to the incompressible Conservation of Mass:

$$\overline{\frac{\partial u_i}{\partial t} + \frac{\partial u_i u_j}{\partial x_j}} = \overline{-\frac{1}{\rho} \frac{\partial p}{\partial x_i} + 2\nu \frac{\partial S_{ij}^d}{\partial x_j}}$$

![](_page_34_Picture_5.jpeg)

![](_page_34_Picture_6.jpeg)

![](_page_35_Picture_0.jpeg)

## LES Equations

$$\frac{\overline{\partial u_i}}{\partial t} + \frac{\overline{\partial u_i u_j}}{\partial x_j} = \overline{-\frac{1}{\rho} \frac{\partial p}{\partial x_i}} + \frac{\overline{2\nu} \frac{\partial S_{ij}^d}{\partial x_j}}{\partial x_j}$$

• Commutative with respect to differentiation:

$$\frac{\partial \overline{u_i}}{\partial t} + \frac{\partial \overline{u_i u_j}}{\partial x_j} = -\frac{1}{\rho} \frac{\partial \overline{p}}{\partial x_i} + 2\nu \frac{\partial \overline{S_{ij}^d}}{\partial x_j}$$

![](_page_35_Picture_6.jpeg)

![](_page_35_Picture_7.jpeg)


### LES Equations

- This filter operation introduces  $\overline{u_i u_j}$  as an unknown
- This is approximated by decomposing the term into:

$$\tau_{ij} = \overline{u_i u_j} - \overline{u_i} \overline{u_j}$$

- $au_{ij}$  is the sub-grid scale (SGS) stress tensor
- The deviatoric SGS stress tensor can be calculated with:

$$\tau_{ij}^d = -2 \, \nu_{SGS} \overline{S_{ij}^d}$$

- Which introduces  $\nu_{sGS}$  as the eddy viscosity or sub-grid scale kinematic viscosity
- An LES model is introduced to solve for  $v_{SGS}$
- The same procedure can be followed for compressible LES but requires the use of Favre filtering







# Favre Filtering

- Key for LES in compressible flows
- Change of variables based on filtered density
- This can be written as:

$$\overline{\rho\Phi}=\bar{\rho}\widetilde{\Phi}$$

• Or more practically:

$$\widetilde{\Phi} = \frac{\overline{\rho \Phi}}{\overline{\rho}}$$





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Constant Smagorinsky Model

• Directly models  $\mu_{SGS}$  based on instantaneous flow state:

 $\mu_{SGS} = \bar{\rho} (C_s \Delta)^2 \left| \tilde{S} \right|$ 

•  $|\tilde{S}|$  is the magnitude of the Favre averaged strain rate tensor:

$$\left|\widetilde{S}\right| = \sqrt{2\widetilde{S_{ij}}\widetilde{S_{ij}}}$$





Constant Smagorinsky Model

- *C<sub>s</sub>* is the Smagorinsky coefficient
  Often chosen to be 0.17
- $\Delta$  represents the element size

$$\Delta = C_P \big( \Delta_x \Delta_y \Delta_z \big)^{1/3}$$

•  $C_P$  factors in finite element solution order P:  $C_P = \frac{1}{P+1}$ 





- Modification of the constant Smagorinsky model
  - Constant model poorly handles laminar and transitional flows
- The Smagorinsky coefficient is now calculated as a function of the instantaneous flow state  $C_s = C_s(x, y, z, t)$
- An explicit filter operation is applied locally
  - Occurs independently of solution or grid filtering

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• This filter is referred to as the test filter





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# Dynamic Smagorinsky Model

- This work examined the performance both filters
  - Sharp Modal Cutoff
  - Laplace
- Test filtered quantity is represented by a hat
- Calculation is based on the Leonard Stress tensor:

$$L_{ij} = \overline{\rho} \widehat{\widetilde{u}_i} \widehat{\widetilde{u}_j} - \frac{\overline{\rho} \widetilde{\widetilde{u}_i} \overline{\rho} \widetilde{\widetilde{u}_j}}{\overline{\rho}}$$

• And the *M<sub>ij</sub>* tensor:

$$M_{ij} = \left(\bar{\rho} |\tilde{S}| \tilde{S}_{ij}^{\vec{d}}\right) - \alpha \hat{\bar{\rho}} \left|\hat{S}\right| \hat{\tilde{S}}_{ij}^{\vec{d}}$$







•  $\alpha$  is the ratio of the grid filter size and the test filter size:

$$\alpha = \begin{pmatrix} \widehat{\underline{\Delta}} \\ \overline{\underline{\Delta}} \end{pmatrix}$$

• For the finite element formulation order is factored in:

$$\alpha = \left(\frac{p_{grid} + 1}{p_{test} + 1}\right)^2$$

• Manipulation of the terms results in:

$$(C_s \Delta)^2 = \frac{1}{2} \frac{L_{ij}^d M_{ij}}{M_{lk} M_{lk}}$$

• Which is then substituted into:

$$\mu_{SGS} = \bar{\rho} (C_s \Delta)^2 \left| \tilde{S} \right|$$





- Both tensors are constructed out of terms:
  - Filtered then assembled
  - Assembled then filtered
- High energy content is associated with high order components of flow
- Leonard Stress tensor becomes zero in smooth flow
  - Results in  $\nu_{\text{SGS}}$  having zero contribution in this flow regime
  - This contrasts with the Constant model that only has zero contribution with zero strain rate

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• Calculated at each integration point, then averaged over the volume:

$$\langle (C_s \Delta)^2 \rangle_e = \frac{\int_{V_e} (C_s \Delta)^2 dV}{V_e}$$

- Then applied on a per element basis
- Clipping was introduced to prevent  $\mu_{SGS}$  from becoming negative

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• If  $\mu + \mu_{SGS} \leq 0$  instabilities will form





Implicit Large Eddy Simulation

- ILES relies strictly on the native viscosity from the Navier-Stokes equations and numerical dissipation arising from the solver
- No small scale physics or structures are captured
- Referred to as *Baseline* or *No Model* in this work

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### Taylor Green Vortex

- Is an unsteady flow that is initially laminar and under goes transition to fully turbulent flow
- Ideal test for SGS models
  - Transition is difficult to handle
- Inherently an incompressible
   problem
  - Mach Number of 0.1 was selected



Contour of Z-vorticitiy at the initial condition [1].





- This is a perodic problem
  - All boundary conditions were perodic
- Domain of  $[-\pi L, \pi L]x[-\pi L, \pi L]x[-\pi L, \pi L]$ 
  - With the characteristic length L = 1.0
- The Characteristic velocity  $U_0 = 0.1$
- The initial density ho=1.0
- Air was chosen as the working fluid
  - $\gamma = 1.4$
  - Pr = 0.71







• Initial Flow field:

$$u = U_0 sin\left(\frac{x}{L}\right) cos\left(\frac{y}{L}\right) cos\left(\frac{z}{L}\right)$$
$$v = U_0 cos\left(\frac{x}{L}\right) sin\left(\frac{y}{L}\right) cos\left(\frac{z}{L}\right)$$

w = 0

$$P = P_0 = \frac{\rho_0 U_0^2}{16} \left( \cos\left(\frac{2x}{L}\right) + \cos\left(\frac{2y}{L}\right) \right) \left( \cos\left(\frac{2z}{L}\right) \right)$$

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#### Taylor Green Vortex

- Initial temperature was considered to be uniform
  - Thus density was calculated by:  $\rho = RT_0$
- The time was normalized with a characteristic convective time:

• 
$$t_c = L/U_0$$
  
•  $\tau = \frac{t}{t_c}$ 







- Key quantities were analyzed to determine the behavior of the TGV simulation runs
- Volume averaged kinetic energy:

$$KE = \frac{1}{\rho\Omega} \int_{\Omega} \rho \frac{u_i u_j}{2} d\Omega$$

• Kinetic energy dissipation rate: dK

$$\epsilon = -\frac{dt}{dt}$$





- Kinetic energy rate is based on the sum of three terms:
  - The  $e_1$  term represents the dissipation arising from viscosity:

$$e_1 = \frac{2}{\rho_0 \Omega} \int_{\Omega} \mu S_{ij}^d S_{ij}^d d\Omega$$

• The  $e_2$  term represents dissipation arising from velocity dilatation:

$$e_2 = \frac{2\mu}{3\rho_0\Omega} \int_{\Omega} (\nabla \cdot u)^2 d\Omega$$

• The  $e_3$  term represents dissipation arising from pressure dilatation:

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$$e_3 = -\frac{1}{\rho_0 \Omega} \int_{\Omega} P(\nabla \cdot u) d\Omega$$

- Due to the incompressible nature of the problem  ${\rm e_2}$  and  ${\rm e_3}$  should be negligible





• Volume averaged turbulent viscosity was also analyzed:

$$\overline{\mu_{SGS}} = \frac{1}{\Omega} \int_{\Omega} \mu_{SGS} d\Omega$$

• The number of degrees of freedom were calculated as:

 $DOF = [(P+1)n_{1D}]^3$ 

• This is based on the solution order P and the number of elements which were constant for each direction

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- Simulations were run in CartDG
- Both the split-form and standard DG discretizations were run
- The explicit fourth-order four stage Runge-Kutaa (RK4) scheme was used for time advancement
  - The 3/8<sup>th</sup> Method was used for the RK coefficients
- CFL = 1.0 was selected





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### Inviscid Taylor Green Vortex

- TGV was run without viscosity
- Fully periodic
- No means of dissipating energy
  TKE should be strictly conserved
- Challenging:
  - Will always be under-resolved for long simulations





#### Results: Inviscid TGV DG

- Total Kinetic Energy
- Run with the standard DG formulation
- Lax-Friedrichs numerical flux scheme
- All cases P2, P3, and P6 filtered to P3 are too dissipative
- Only the lowest order P2 case, shown in purple, was stable and ran to completion







#### Results: Inviscid TGV DG

- P3, shown in green, reaches approximately  $\tau \approx 16.0$  before crashing
- Adding in filtering from P6 to P3, shown in red, destabilized the solution further
  - Due to worsening polynomial aliasing errors
  - The quadrature points are as if the solution was P6



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### Results: Inviscid TGV Split-form

- Total Kinetic Energy
- Run with the split-form
- No numerical flux dissipation was introduced
- P2, shown in purple, runs to completion
  - Negligible changes in KE
- P3, shown in cyan, runs to completion
  - KE begins to decrease at  $\tau \approx 13.0$
  - Then begins to rise after au pprox 18.0



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#### Results: Inviscid TGV Split-form

- Both P4 cases crash
- The case with 14 elements per direction shown in yellow
  - Preserves KE until  $\tau \approx 7.5$
  - At this point it decreases
  - It begins to increase at  $\tau \approx 10.0$  before crashing
- The case with 16 elements per direction
  - Preserves KE for its entire life before crashing



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- Run with viscosity
- Reynolds Number was varied to examine the effects
  - RE=1,600
  - RE=20,000
- Run with and without LES models
  - Baseline will refer to ILES





#### Results: TGV DG

- Run with viscosity
- Reynolds Number was varied to examine the effects
  - RE=1,600
  - RE=20,000
- Run with and without LES models
  - Baseline will refer to ILES





### Results: TGV DG vs Split-Form

- TKE with Re-20000 P6

  - DNS is in Purple
    Standard DG formulation in green
  - Split-form in red
- The standard DG formulation is unstable
  - In the regime before the instability is more accurate then the splitform



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### Results: TGV DG vs Split-Form

- The split-form under predicts compared to the DNS
  - The transition is predicted earlier



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## Results: TGV Split-form LES

- Total Kinetic Energy at Re=1,600
  - DNS results are in purple
  - Baseline simulation with no LES model is shown in Blue
  - The simulation with Constant Smagorinsky model model is green
  - The simulation with Dynamic Smagorinsky model with the modal cutoff filter set to P4 is in orange
  - The simulation with Dynamic Smagorinsky model with the Laplacian filter is in red



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## Results: TGV Split-form

- All simulations
  - under predict KE compared to the DNS
  - predict transition earlier than the DNS
- The Constant Smagorinsky Model is the most under predictive
  - It predicts transition the earliest
- The Dynamic Smagorinsky Model with the modal cutoff filter is slightly more dissipative than the baseline, the splitform with no model
- The Dynamic Smagorinsky Model with the Laplace Filter is nearly identical to the baseline as the model has very little contribution



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### Results: TGV Split-form

- Energy dissipation rate at Re=1,600
  - DNS results are in purple
  - Baseline simulation with no LES model is shown in Blue
  - The simulation with Constant Smagorinsky model is green
  - The simulation with Dynamic Smagorinsky model with the modal cutoff filter set to P4 is in orange
  - The simulation with Dynamic Smagorinsky model with the Laplacian filter is in red



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#### Results: TGV Split-form

- All simulations over predict energy dissipation and predict transition earlier than the DNS
- The Constant Smagorinsky model is the most dissipative until  $\tau \approx 7.5$ 
  - After this time it is the least dissipative









### Results: TGV Split-form

- The Dynamic Smagorinsky model with the cutoff filter has close performance with the baseline
  - At  $\tau \approx 5$  it is slightly more dissipative, this is responsible for the divergence seen in the figure of TKE shown previously
- The Dynamic Smagorinsky model with the Laplace filter has too small of a contribution and produces negligible differences when compared to the baseline simulation



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## Results: TGV Split-form

- Modeled eddy viscosity is shown
  - This is volume averaged over the whole domain
- The Constant Smagorinsky Model results are shown in purple
- The Dynamic Smagorinsky Model results with the modal cutoff filter is shown in green
- They Dynamic Smagorinsky Model results with the Laplace filter is shown in red



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## Results: TGV Split-form

- The Constant Smagorinsky model predicts the largest  $\mu_t$  during the entire simulation
- The Dynamic Smagorinsky model with the cutoff filter predicts a  $\mu_t \approx 0$  until  $\tau \approx 2.5$ 
  - This is when the flow is Laminar
- The Dynamic Smagorinsky model with the Laplace filter predicts  $\mu_t \approx 0$  for the lifetime of the simulation
  - A closer examination shows similar behavior to the modal cutoff filter
  - The magnitude is too small to have an effect on the flow



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## Results: TGV Split-form High Re

- Total Kinetic Energy at Re=20,000
  - DNS results are in purple
  - Baseline simulation with no LES model is shown in Blue
  - The simulation with Constant Smagorinsky model is green
  - The simulation with Dynamic Smagorinsky model with the Laplacian filter is in orange
  - The simulation with Dynamic Smagorinsky model with the modal cutoff filter set to P4 is in red



Split: Re20000 P6N16 = 112^3 dofs





## Results: TGV Split-form High Re

- Again, all cases under predict KE and predict transition earlier than the DNS
- The Constant Smagorinsky Model predicts the transition the earliest
  - After transition it over predicts KE






## Results: TGV Split-form High Re

- The Dynamic Smagorinsky with the Laplace filter is notably different than the baseline simulation with no LES model
  - Improvement over the lower Reynolds Number case
  - It predicts a higher KE than the baseline
  - Transition occurs at nearly the same time as the baseline
- The Dynamic Smagorinsky with the cutoff filter predicts transition closer to the DNS results than all other cases







# Results: TGV Split-form High Re

- Energy dissipation rate at Re=20,000
  - DNS results are in purple
  - Baseline simulation with no LES model is shown in Blue
  - The simulation with Constant Smagorinsky model is green
  - The simulation with Dynamic Smagorinsky model with the Laplacian filter is in orange
  - The simulation with Dynamic Smagorinsky model with the modal cutoff filter set to P4 is in red







## Results: TGV Split-form High Re

- All cases predict a higher dissipation rate before transition
  - $\tau \approx 7.5$
- After this point they under predict the energy dissipation rate until  $\tau \approx 13.0$
- The DNS indicates there is a secondary peak at  $\tau \approx 12.0$ 
  - This is not seen in the lower Re case
  - All LES cases capture this







# Results: TGV Split-form High Re

- The Constant Smagorinsky Model predicts the onset of transition the earliest
  - After  $\tau \approx 6.0$  it predicts the lowest dissipation rate of the LES cases
- The Dynamic Smagorinsky Model case with the cutoff behaves similarly to the baseline no model case until  $\tau \approx 6.0$ 
  - This case also has a significant amount of fluctuations after  $\tau \approx 6.0$







### Results: TGV Split-form High Re

- The Dynamic Smagorinsky Model case with the Laplace filter behaves similarly to the baseline no model case until  $\tau \approx 9.0$ 
  - Unlike the Low Re case this filter has a significant contribution
  - In many areas this case performs closer to the DNS results than the other LES cases





# Results: TGV Split-form High Re

- Modeled eddy viscosity is shown
  - This is volume averaged over the whole domain
- The Constant Smagorinsky Model results are shown in purple
- The Dynamic Smagorinsky Model results with the modal cutoff filter is shown in green
- They Dynamic Smagorinsky Model results with the Laplace filter is shown in red



Split: Re20000 P6N16 = 112^3 dofs



## Results: TGV Split-form High Re

- As seen in the low Re case the Constant Smagorinsky model predicts the largest  $\mu_t$  during the entire simulation
- The Dynamic Smagorinsky model case with the modal cutoff filter predicts a near zero  $\mu_t$  until about  $\tau \approx 4.0$ 
  - This is desirable as the flow is laminar early in the simulation
- The Dynamic Smagorinsky model case with the Laplace filter again has the lowest volume averaged  $\mu_t$ 
  - Unlike the low Re Case  $\mu_t$  has a notable contribution on the flow



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#### Conclusions

- The standard DG discretization is unstable for the inviscid lGV
  - It is unstable in every case
- The split-form discretization can preserve KE
  - This is seen in the inviscid TGV
    - No dissipation was added with a numerical flux scheme
    - Some cases were stable for the whole simulation life with negligible loss in KE
- The standard DG discretization is unstable when the split-form discretization is stable
  - When the stand DG discretization is stable it is less dissaptive than the split-form discretization







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#### Conclusions

- At Re=1600 all LES models under predict KE when compared to the DNS results
  - The case with no LES model was the least dissipative
- At Re=20000 all LES models under predict KE when compared to the DNS results
  - The Dynamic Smagorinsky model cases had less dissipation than the no model LES case
    - This arises from the extra dissipation from the model is applied in key flow areas





### Conclusions

- The Constant Smagorinsky Model was the most dissipative when compared to the Dynamic Smagorinsky model and no model cases
  - This model has poor performance for laminar flows
- The Dynamic Smagorinsky model cases performed better than the Constant Smagorinsky Model case
  - This model has better performance in laminar flows
  - Accurately capturing the early laminar flow and transition leads to better accuracy





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### Future Work

- The split-form discretization chosen was kinetic energy preserving
  - Other discretizations exist with different variable formulations or preserved properties
- Other LES models should be explored
  - Dynamic Heinz
  - Wall-Adapting Local Eddy-Viscosity (WALE) model

- Other turbulent flow problems should be analyzed
  - Turbulent channel flow





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Questions









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